



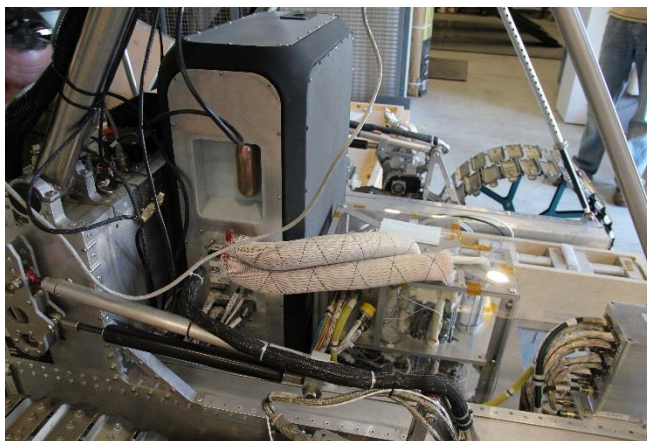
OVEN & LAVA Subsystems in the RESOLVE Payload for Resource Prospector

OVEN – JSC

LAVA – KSC with JPL

1.2 RESOURCE PROSPECTOR SHALL BE CAPABLE OF OBTAINING KNOWLEDGE ABOUT THE LUNAR SURFACE AND SUBSURFACE VOLATILES AND MATERIALS

- Full Success Criteria: Take **both sub-surface measurements of volatile constituents via excavation and processing** and surface measurements, at multiple locations
- Minimum Success Criteria: Take either **sub-surface measurements of volatile constituents via excavation and processing** or surface measurements, at multiple locations



*RESOLVE Payload Hardware
for Field Demonstration in 2012*

Processing & Analysis

Oxygen & Volatile Extraction Node (OVEN)

- Volatile Content Extraction by warming
- Total sample mass

Lunar Advanced Volatile Analysis (LAVA)

- Analytical volatile identification and quantification in delivered sample with GC/MS
- Measure water content of regolith at 0.5% (weight) or greater
- Characterize volatiles of interest below 70 AMU



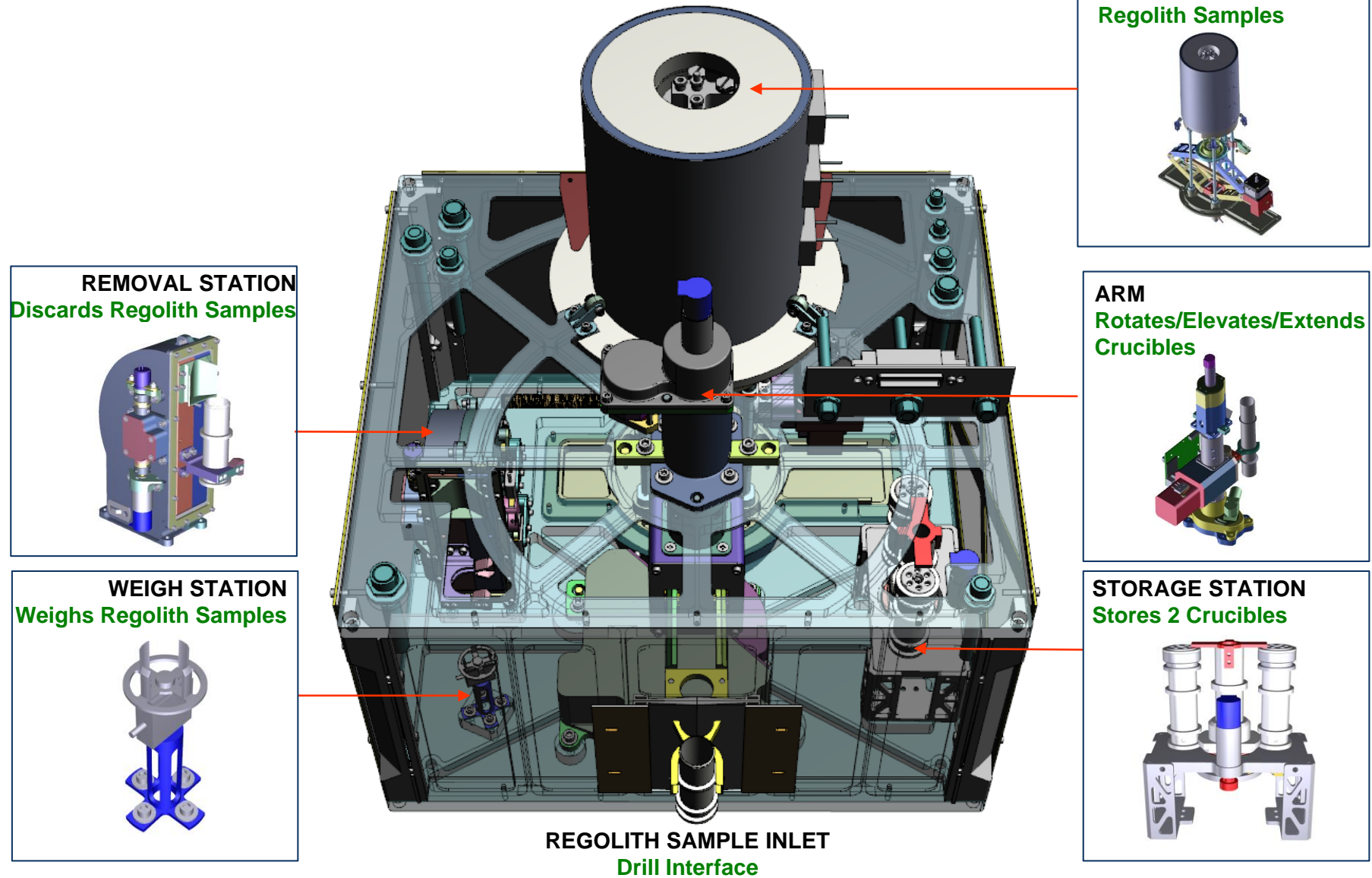
Oxygen & Volatile Extraction Node (OVEN)



Multiple functions

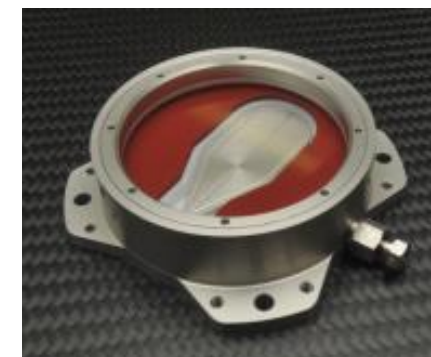
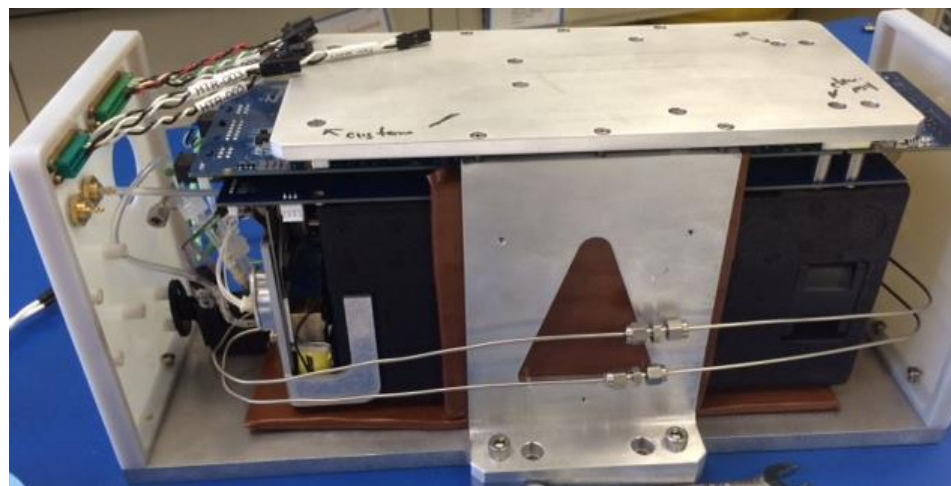
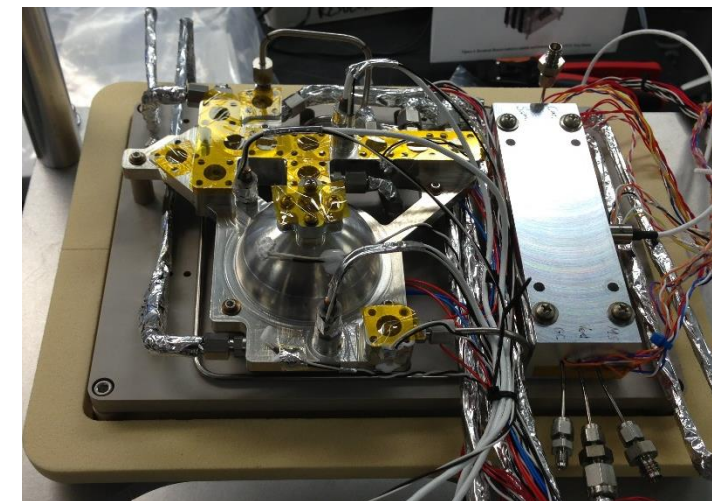
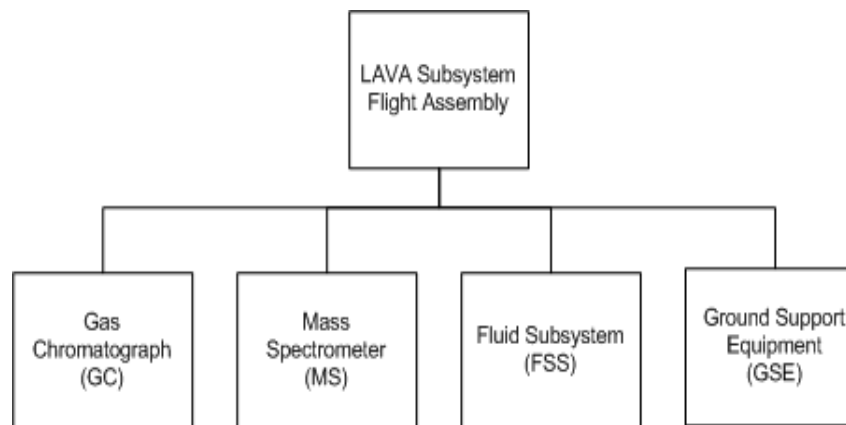
- Receive sample from drill
- Weigh sample pre and post processing
- Heat sample, build pressure from volatiles
- Transfer volatile sample to LAVA Subsystem
- Discard sample

OVEN Demonstration Unit for RP-15



Lunar Advanced Volatile Analysis (LAVA)

- Purpose: Identify and quantify water as well as other low molecular weight species of interest to ISRU and Science community
 - Volatiles are stored in Surge Tank where the Pressure & Temperature are measured, gas sample analyzed by GC-MS to identify constituents.
 - Gases of interest are H_2O , CO , CO_2 , H_2 , H_2S , NH_3 , SO_2 , CH_4 , C_2H_4 .
 - Water that is evolved will be condensed and photographed, demonstration of resource storage.



LAVA Physical Architecture - Flight

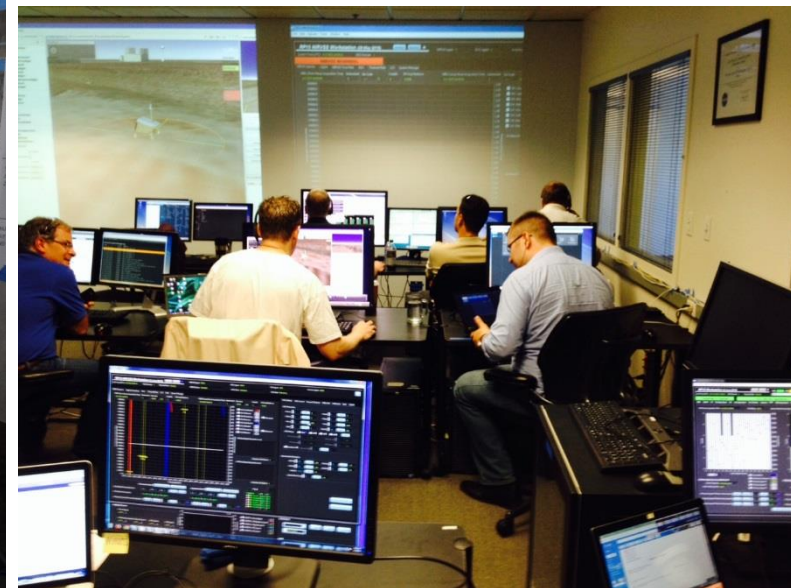
- Automated operations using Virtual Machine Language (VML), which has been used in previous flight missions
- Using flight-forward payload and ground software with our development hardware allowed for very early software validation, operator training, remote instrument integration and distributed control capability
- Distributed ops between labs, integrated payload hardware, and operational tests against system models are used throughout the payload development



KSC Firing Room 4 Control Room



Payload Integrated Test on Rover at JSC

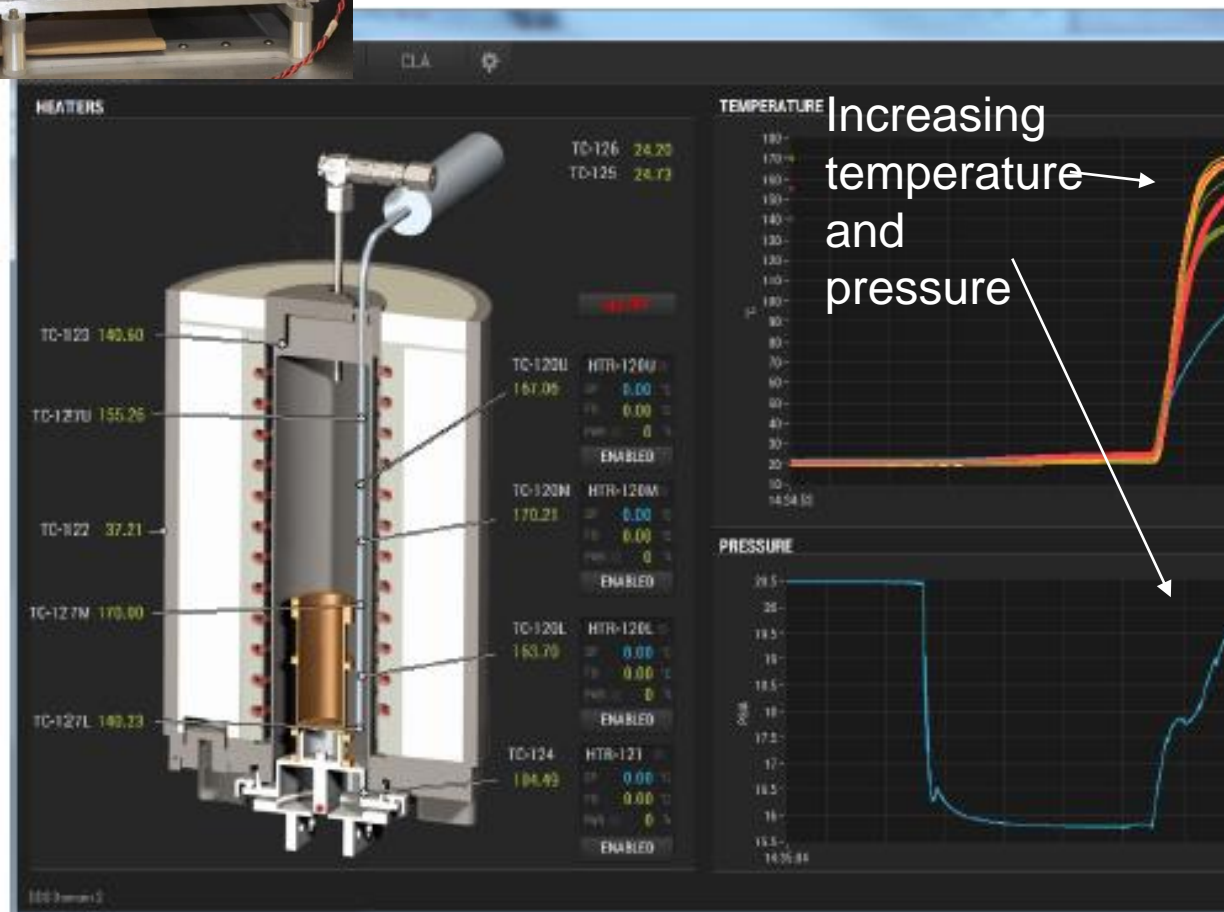
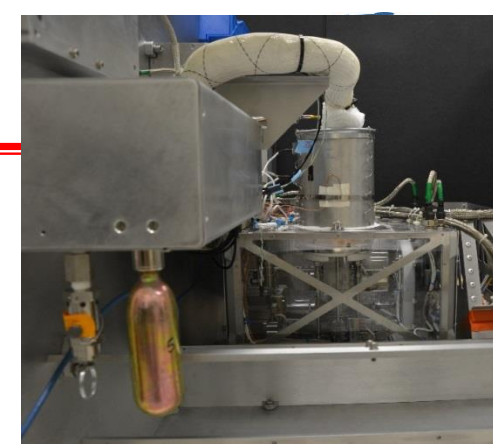
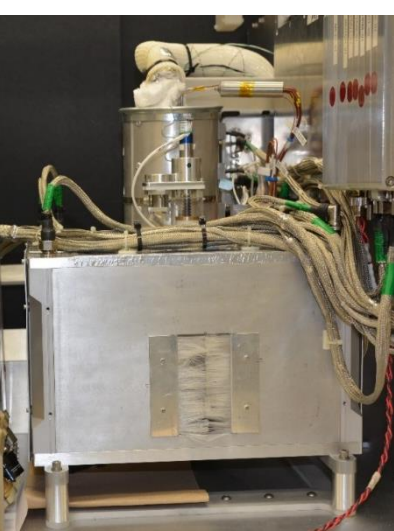


ARC Control Room

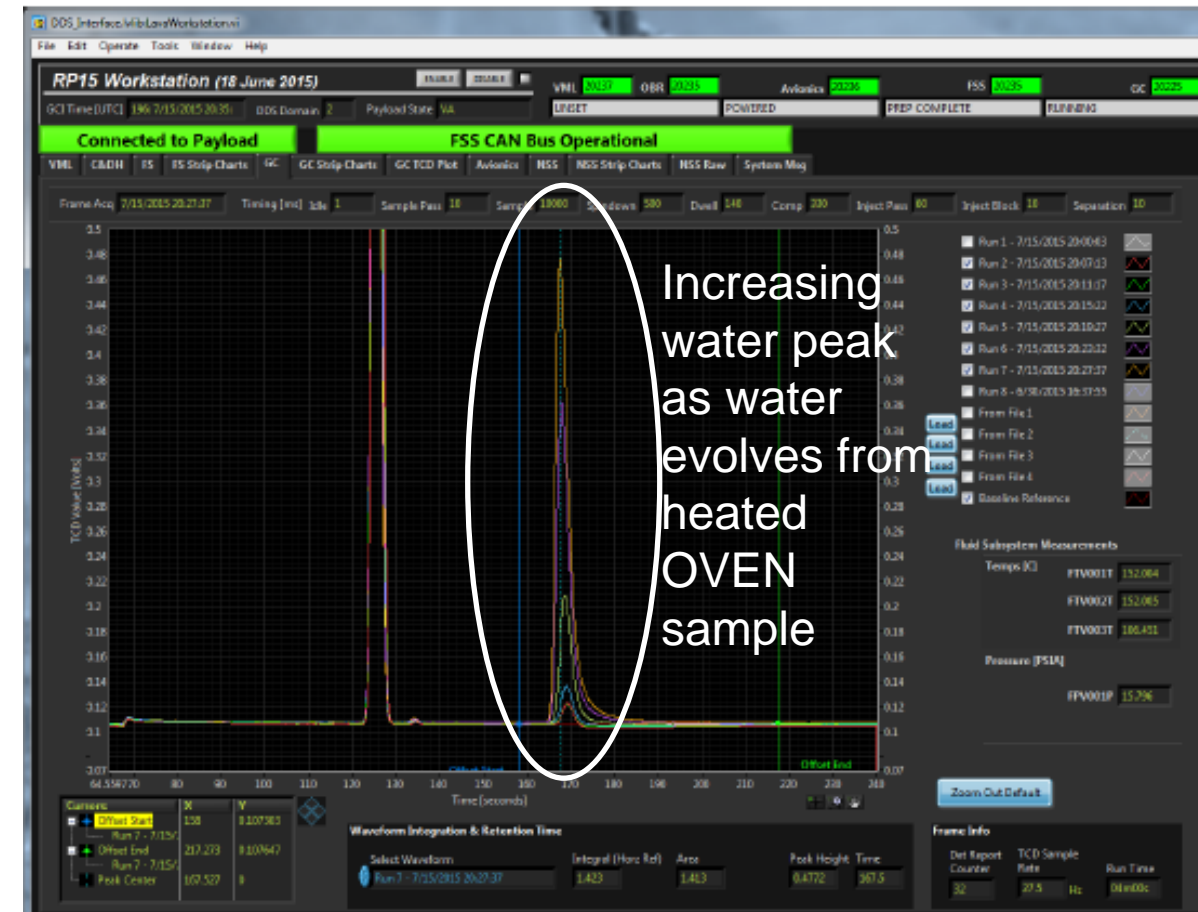
OVEN-LAVA Data for RP-15 Volatile Analysis

RP-15 Limited instrument applicability due to atmosphere

Volatile analysis demonstration analyzing measured increasing water concentration as simulant sample temperature increases



OVEN User Interface



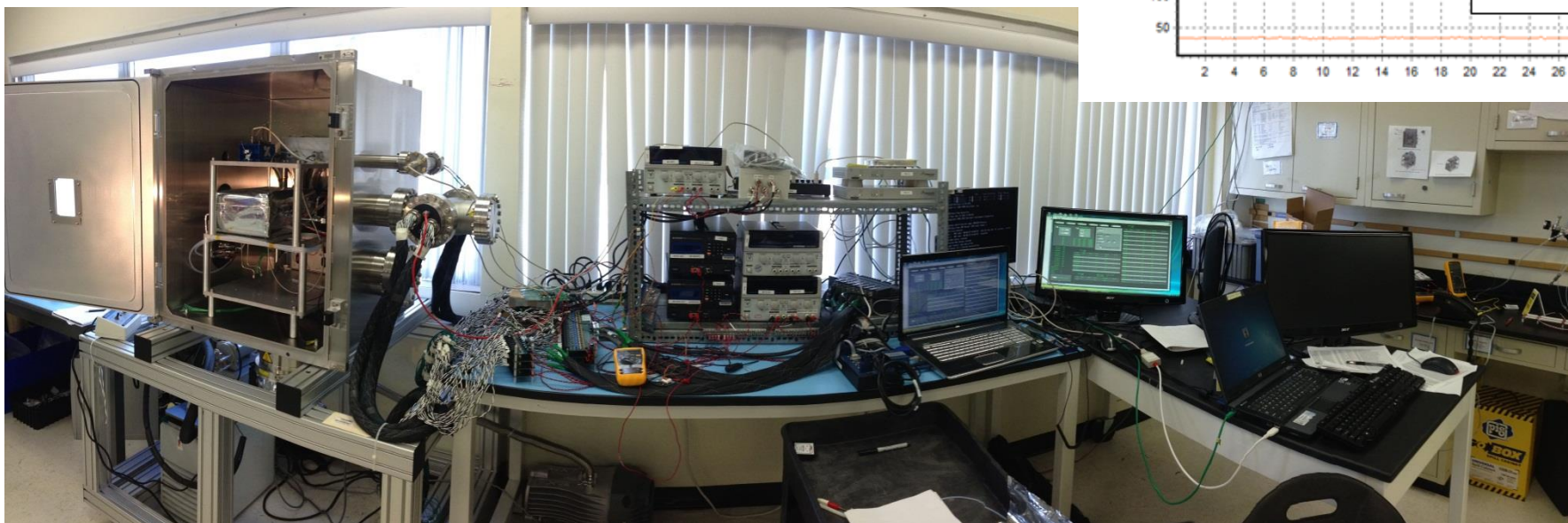
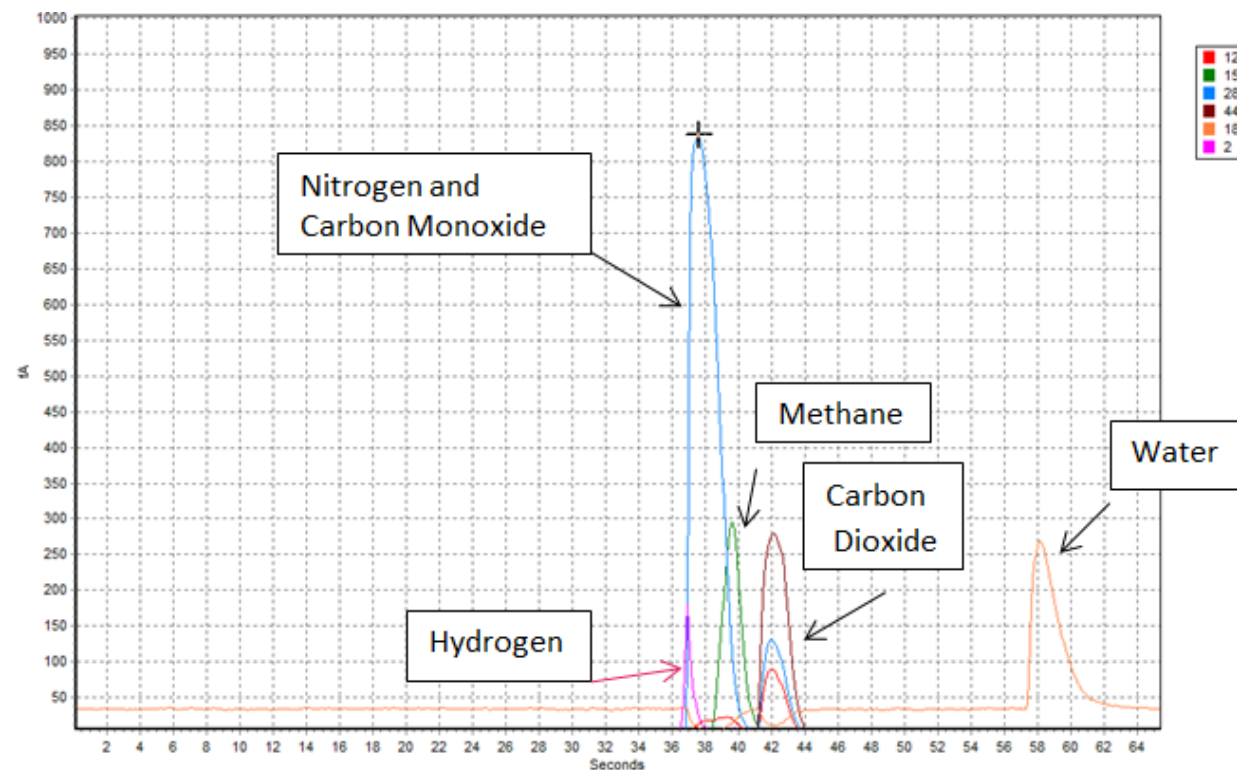
LAVA GC User Interface



Flight forward design – modified COTS



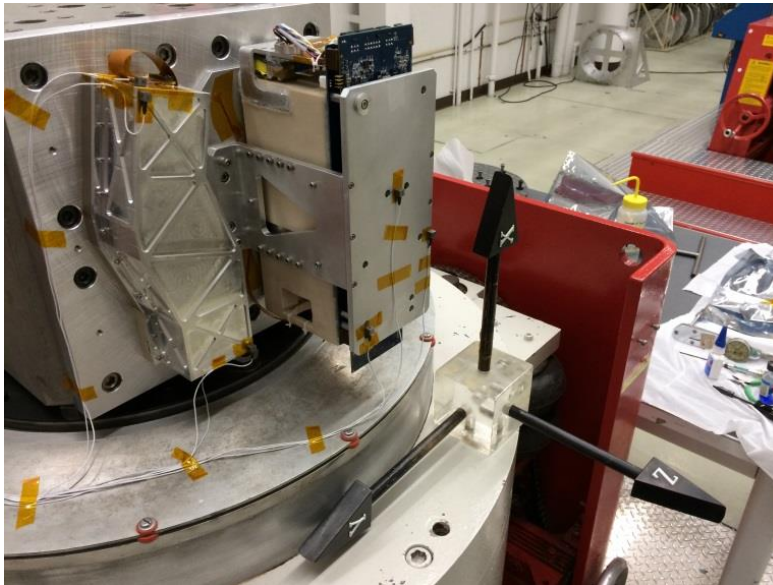
- Utilize components from other missions where possible within schedule/cost
- Raise TRL by testing in thermal vacuum chamber (more flight like for subsystem until integrated payload and rover tests in vacuum)
- Integrated LAVA hardware testing with Payload Software and Avionics at KSC, GC-MS at JPL



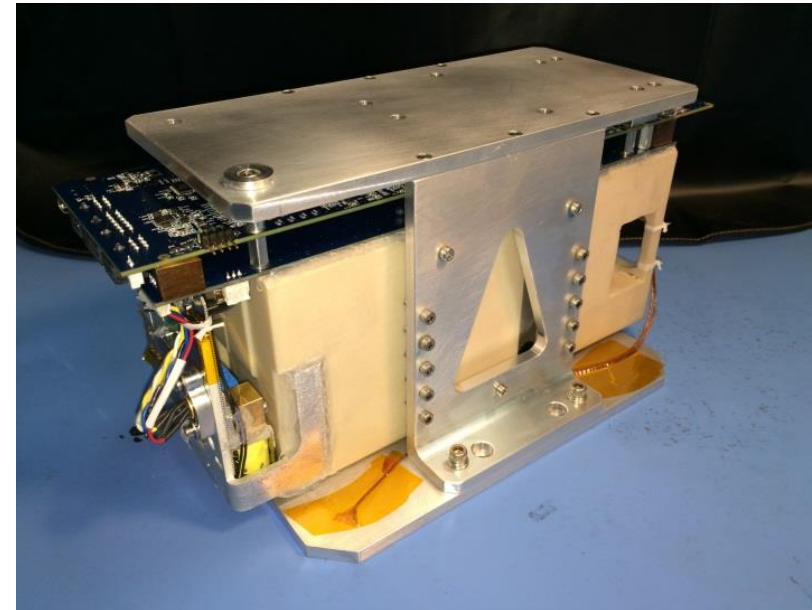
Modified COTS Vibration Testing

- Random Vibration testing (FY14):
 - Completed design/construction of ‘modified COTS’ GC.
 - Completed vibe testing of mass spec and GC at JPL.
 - Both mass spec and GC pass post-vibe limited functional tests.
 - Now TRL6 mass spec: engineering model tested in relevant environment. Design essentially locked.
 - **TRL5/6** GC: “brassboard +” tested in relevant environment, but design changes likely.

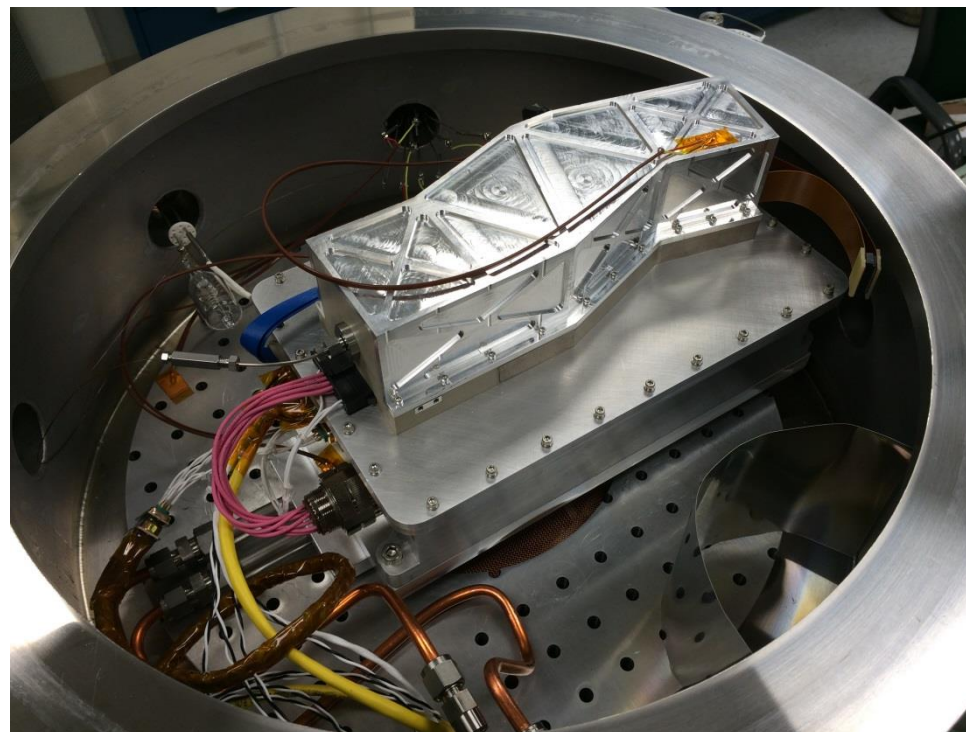
3-axis 14.1grms random vibe at JPL



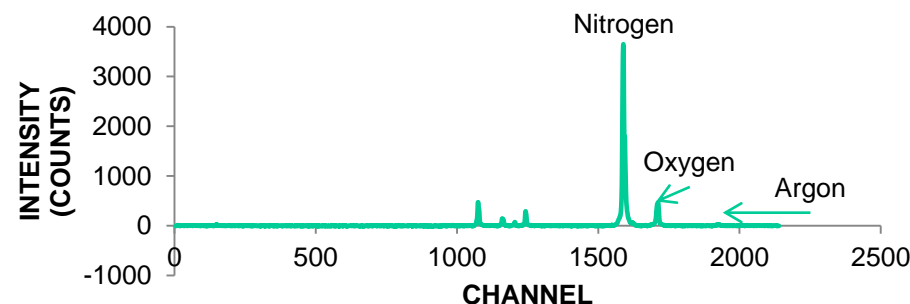
Modified COTS GC



- Accomplishments:
- VDU Mass Spectrometer:
 - Integrated electronics and mass spec into bell jar. MILESTONE
 - Test data produced: Verified operation in vacuum of system
 - MS is TRL6, Electronics TRL5 [needs vibe test], System is TRL5.
- Low-mass electronics enclosure:
 - Layup tool being fabricated.
 - Design being analyzed.
 - Expect ~40% reduction in mass of final assembly using the pre-preg carbon fiber material vs aluminum.
- Plans:
 - Complete fabrication of layup tool (June) & procure material
 - Lay up of carbon composite electronics enclosure (July-August)
 - Gather and repeat mass spec test data (June – September)



RP20 Mass spectrometer with electronics in vacuum chamber, ready for testing.



Mass spectrum of air measured using TRL5 mass spectrometer system.



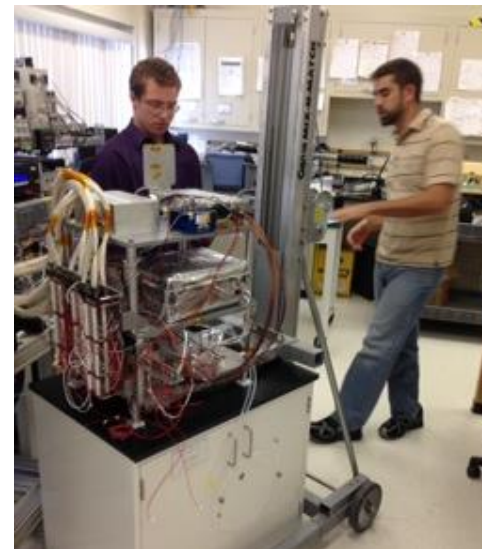
Current Status and Future Work



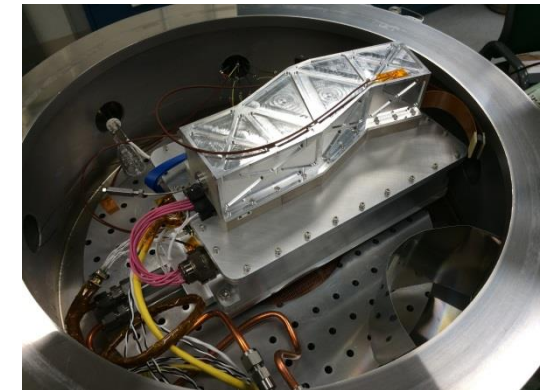
- Integrated LAVA-OVEN test of flight-like design tested in thermal vacuum chamber that includes OVEN reactor and LAVA fluid subsystem manifold, water droplet demonstration, gas chromatograph and mass spectrometer
- Two COTS mass spectrometers under evaluation that have data supporting proposed requirements for volatile analysis
- Employing modified COTS, working with vendors closely for instrument and component testing for evaluation and testing
- Working with Science PI/ISRU community to understand analysis priorities and ensure payload can meet quantitation goals



*Collaboration with JPL, SBIR companies,
and instrument vendors*



Integrated Stack ready for install in Vac Chamber



MS system configured in environmental chamber